



EXAMINATION OF FATIGUE BREAK SURFACES IN ALUMINUM ALLOY 2024 T351

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ABSTRACT

A way for measuring of fracture surfaces is planned. the strategy consists of determinant the abstraction distribution of serious fractographic options victimisation the Scanning microscopy. The results obtained below constant amplitude fatigue tests show that completely different crack propagation will be known by this method. important effects because of load quantitative relation changes are quantified. The technique developed is then applied to fracture surfaces obtained below variable amplitude loading. The analyses of the results bring out variations in crack growth mechanisms below the studied check conditions.

1. INTRODUCTION

TO characterize the behavior of a fabric in fatigue propagation, Paris law [1] is usually used. This law relates the crack rate da/dN and therefore the cyclic amplitude stress intensity issue ΔK . This relationship will show many stages separated by transitions [2–4]. In predominant the small structural size, the close surroundings and therefore the frequency of solicitation will modify the behavior of the crack growth curves (da/dN vs. ΔK) and therefore the transitions [3]. a full of life approach [5, 6] supported the theoretical model of Weertman [6] permits a far better comprehension of mechanisms of propagation with a attainable correlation between megascopic and microscopic mechanisms. However, once fatigue failures occur, engineers would like tools to scan the fracture Surfaces. the most problem, that is encountered with this respect, is to search out an instantaneous correlation between the fracture topography and therefore the stress operational on the structure. As our data and understanding of fracture phenomena improves, thus will the quantity of knowledge, that we are able to acquire from a failure analysis of metallike elements. within the past, AN investigator sometimes was happy with distinctive the sort of failure, the locating fracture origin, determinant the fabric compositions, microstructure and properties and providing a general clarification for the failure [7]. Today, with the event of scanning microscopy, that has become an essential tool for the analysis of fracture surfaces, it's attainable to spot the various fractographic options representative of fatigue failures, such fatigue striations, crystallographic aspects or dimples [8,9]. Ever since the observations of Forsyth [10] regarding the formation of fatigue striations, completely different authors have shown that there's an honest correlation between striation spacing and megascopic crack rate [11]. Since the origination of the scanning microscope (SEM), researchers within the fracture of metals have tried quantitative measurements on scanning microscope fractographs in a technique or another. In recent years, many major contributions to the measuring of fracture surfaces have centered on varied aspects of the matter. This paper presents the results of a way of quantification of fracture surfaces ensuing from constant amplitude cyclic tests [12] and it's applied to samples of variable amplitude loading conditions. The results analyzed with relation to completely different parameters governing the crack propagation.

2. BEXPERIMENTAL CONDITION

A. Material

the fabric thought of during this study may be a high strength aluminium alloy 2024 T351. Tables I and II offers the nominal composition and mechanical properties of the fabric.

B. 5BTest conditions

Constant amplitude (CA) crevice growth tests were conducted victimisation AN Instron servo-hydraulic machine at 5 R ratios of zero.01, 0.1, 0.33, 0.54, and 0.70.

The variable amplitude (VA) loading tests were derived from a applied mathematics load history characterizing the loading conditions on a transport craft wing [13]. The loading sort and therefore the completely different load levels ar shown in Fig. 1 and Table III severally.

After the tests, the broken surfaces were examined below scanning magnifier, at {different|totally completely different|completely different} magnifications variable from two hundred to ten thousand to spot and quantify different

fractographic options. a way for measuring of fracture surfaces is planned, the subsequent technique technique conferred in [13, 14] is employed. within the gift study the relative quantity on the most fractographic feature ar: - Pseudo-cleavage aspects or fabric Patterns [HB] These aspects are crystallographic in nature and owe their name to the everyday fabric form (see fig. 2b). These options are referred to as « fish-bone structure » that is encountered at low K values this structure differs from the cleavage by the aspect that crack path differs slightly from the outlined crystallographic planes fig.2b. -Striations one (S1), that ar classical ductile striations [15] which might be related to the megascopic crack rate. -Striations a pair of (S2), they're additional pronounced markings than striations one and therefore the spacing between these striations appear to be freelance of the megascopic crack growth rate(fig.2a) - Dimples one (D1) that represent decohesions determined at low K values (fig.2b) -Dimples a pair of (D2) that ar classical dimples determined at comparatively high K values (fig.2a).

3. EXPERIMENTAL RESULTS

A.Result below Constant Amplitude Loading

The crack curve showing da/dN as a operate of ΔK is given in Fig. 3. The results ar adore those obtained by Wanhill [3] for similar material. we have a tendency to noted that within the Fig. 3, the existence of transitions (T1, T2, T3) characterised by a amendment of slope on the curves once the relation da/dN vs ΔK is completely different. the various transitions determined ar known in Table three.

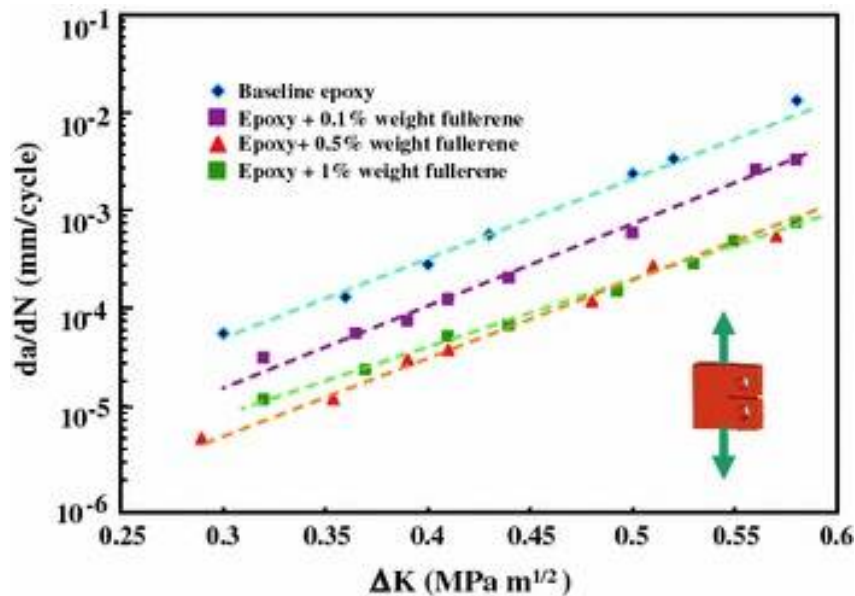


Fig. 1 Evolution of the crack growth rate da/dN with respect to the amplitude stress intensity factor ΔK

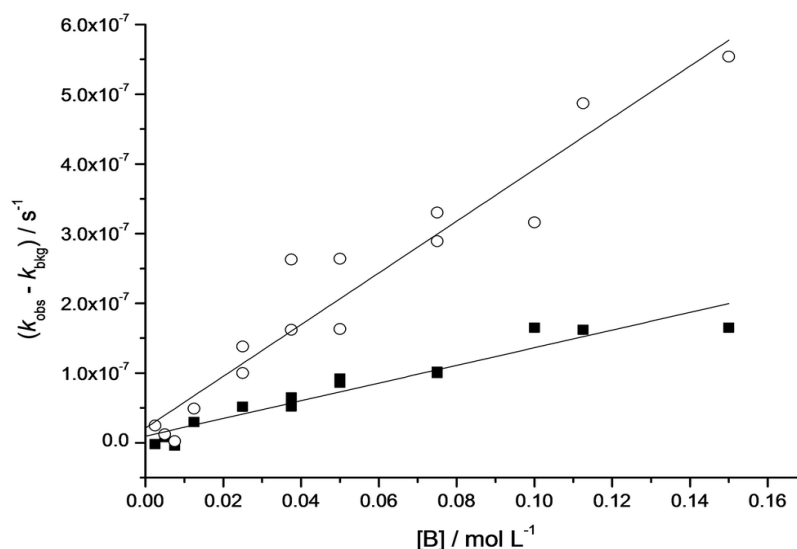


Fig. 2 Evolution of pseudo-cleavage features at $R=0.33$



In a initial approach, solely 2 categories of options were considered: a primary one together with the pseudo-cleavage areas and striations and a second together with D1 and D2 dimples. Typical results are given in Fig. four for $R = 0.33$. It will be noticed that at low K_{max} values, nearly the whole surface is occupied by pseudo-cleavage aspects. At $K_{max} = 9 \text{ MPa m}^{1/2}$ values above T_1 transition, dimples seem and their relative share will increase with K_{max} till reaching a most.

The variations with respect of the R quantitative relation are visible within the evolution of the dimples D2. As will be determined in Figures five and six at the lower R price the relative share of dimples is far below that of the striations. within the middle K_{max} vary a tableland is determined at regarding two hundredth. At high K_{max} values the sharp visit the striated space is in the course of a pointy increase within the mark space. At $R = 0.70$ it will determined that the relative areas occupied by the 2 main fractographic options are nearly a similar within the middle K_{max} vary. Finally, the sharp decrease in striations is additionally related to a rise in dimples at low R values. Striations are shaped as results of slip mechanism and their relative spacing is associated to ΔK [9, 16].

4. CONCLUSION

The evolution of the everyday fractographic options has been related to the various loading regimes that are separated by characteristic transitions. Moreover, the influence of load quantitative relation R and most stress intensity issue K_{max} is taken under consideration by the analysis technique. On the opposite hand, an exceptional relationship between da/dN and ΔK will be given consistent with crack growth regimes. The relation takes under consideration the distinction in behaviour of mechanisms governing the fatigue cracking. For the variable amplitude loading thought of during this study, the fractographic analysis recommends the conception of a similar R_{eff} load quantitative relation primarily based upon injury accumulation principle. once the crack advance results from a striation mechanism R_{eff} corresponds with the R price of the very best loading level, though it is analogous to GAG (Ground Air Ground) cycle for crack growth rates below $5 \times 10^{-6} \text{ m/flight}$.

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